

FIG. 2 (MRB PROGRAM MANAGER)

61-13		11200	5	20	202 202 411 401	
	Check SECONDARY LOW-PITCH STOP (BETA) operation. O	400FH	- i	6		rovise i
61-13	Check SECONDAKY LOW-PIICH SIOP (BEIA) operation. U	400rn	<u>-</u>	5	251, 222, 411, 421	20.
71-10	Verify DRAIN COLLECTOR TANK RETURN LINE check valves operation. O	ე	0.1	06	433	revise
71-12	Check EEC Nh OVERSPEED PROTECTIVE FUNCTION OPERATION (MSI 73-20-02). O	600FН	0.45	OP	221	revise
760-07	Check flight idle stop system. O	1	0.08	0b	221, 222	revise
79-10	Check engine model PW118/118A/118B chip detector. NOTE -	SPECIAL	0.17	90	412	revise
	According to manufacturer's recommendation approved by Canadian DOT. (See MRB Appendix 8.7 or MPG Appendix VI). O				·	
	UPPER FUSELAGE ABOVE CABIN FLOOR	V FLOOR				
24-08	Test MAIN BATTERY load capacity. O	5A	6.0	FC	115	revise
24-09	Check MAIN BATTERY electrolite level. 1	4	0.5	<u>N</u>	115	revise
0-131-02E	5310-131-02E Toilet flush valve support. NOTE - Applicable to FWD lavatory installation only. 1	۲٧	0.08	ESU	131, 132	revise
O-143-01E	2710-143-01E Perform external surveillance in wing center section. Lower skin panel and access panels from CL to wing rib 3. Requires fairing removal. 1	ZA	Ю	ESU	143	revise
	TAILCONE & EMPENNAGE GROUP	ROUP				
322-01Z	20NE 322 - DORSAL FIN - Perform external surveillance of the	₹	0.08	nsı	322	revise
	one, including visual examination of the vortex generators (Fre-Mod. SB 120-055-0008). 1				-	
323-01Z	ZONE 323 - VERTICAL STABILIZER LEADING EDGE - Perform	14	0.08	ESU	323	revise
	external surveillance of the zone, including visual examination of the vortex generators ($Pre-MOD$. SB 120-055-0008). 3					
326-012	20NE 326 - RUDDER I - Perform external surveillance of	14	0.17	nsa	326	revise
78	987	<i>88</i> ,	ક	265	% %,	ī

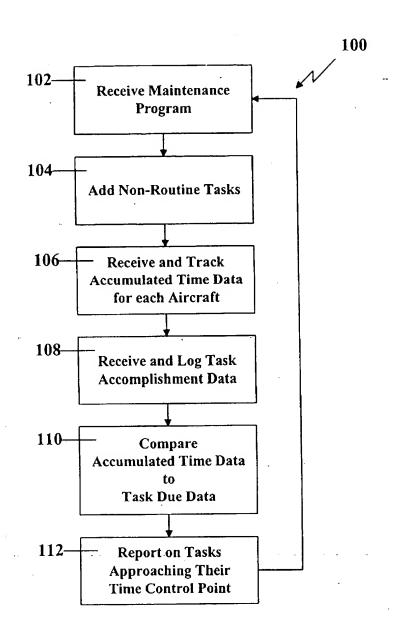


FIG. 5 (TRACKING MANAGER)

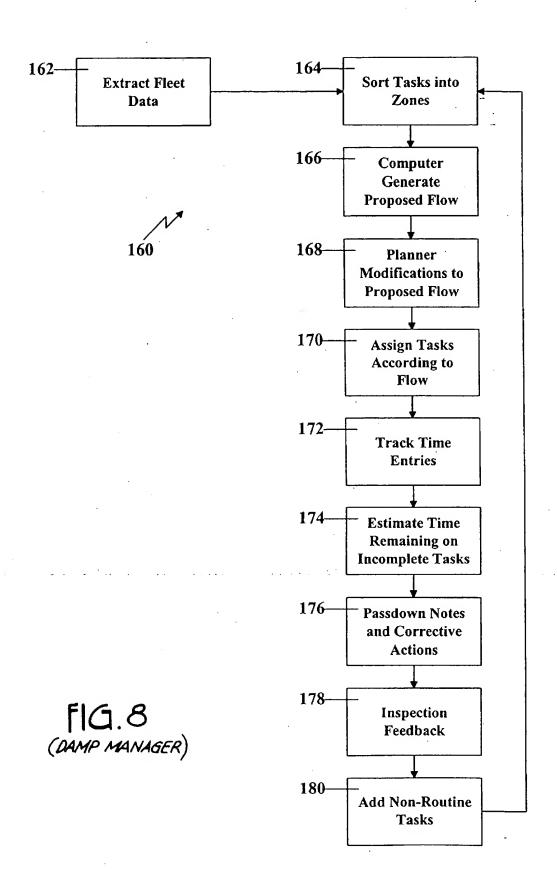
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r. Sinex				VIEW		VIEW	VIEW	VIEW	VIEW	VIEW	
Session User: Sinex	Permissions 🔽		DUE	dte	24268 cyc	dte hrs 47794 cyc	dte hrs 31789 cyć	dte hrs 32000 cyc	dte hrs 24986 cyc.	dte . hrs 27789 cyc	(130 132
	Parts 0		DONE	8/23/99 dte 22871.1 hrs	23468 cyc	9/26/00 dte 23160.1 hrs 23794 cyc	10/25/99 dte 23152.1 hrs 23789 cyc	dte hrs O cyc	10/2/99 dte 23145 hrs 23786 cyc	10/25/99 dte 23152.1 hrs 23789 cyc	821/
	Administration o Maintenance o Line Maint o Program Mgt o	STRUCTURES	DESCRIPTION	5210-811-01 Passenger/crew entry door visible structure, stops and fittings.	Program Interval A2, A5, A11, A17, A8, A14, A20,	5210-811-011 Passenger/crew entry door internal elements, including stops and fittings. Requires door lining removal. Program Interval 24000FC,	5220-812-01E Escape hatches visible structure. Program Interval 8000FC,	5220-812-01E Escape hatch internal elements, including window frame. Requires door lining removal. Program Interval 32000FC,	5220-812-02E Perform external surveillance in escape hatch window frame. Program Interval A3, A8, A13, A18,	5230-813-01E Cargo door visible structure. Program interval 4000FC	98/5
Aircraft: TEST	General 🗸 🗚	,	SOURCE	5210-811-01E Interval	800 FC	5210-811-011 Interval 24000FC	5220-812-01E Interval 24000FC	5220-812-01E Interval 8000FC	5220-812-011 Interval 32000FC	5220-812-02E Interval 1200FC	421
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	55 \D				Limits	Cyc		•					
	Parts n 10 cycles	korders			154 L	Hrs	1200	750	750	1600	1200		400
	Program Mgt 🔽 O hours or within 1	Edit Workorders		korder	leted	Date			1/22/00			11/24/00	10/2/00
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	Line Maint O	-146		Sele	152 La	Hrs	12457.0	13050.0	13050.8	12357.0	13050.6		19965.5
	0/2000		11/30/2000	748		Meter	APU	APU	APU	APU	APU	A/C	A/C
Aircraft: N205CA	General O Administration O Maintenance O Line Maint O Program Mgt O Parts Tasks Due by 11/30/2000 or within 1000 hours or within 10 cycles	142) Select tasks due within	1000 Hours or 10 Cycles or by	750 Retrieve Records	Description	Description	24-10 Restore APU STARIER-GENERATOR. NOTE - According to manufacturer's recommendation. (See MRB Appendix B.7 or MPG Appendix VI).	49-06.Discard GARRET APU model GTCP36-150(A)(AA) OIL FILTER ELEMENT. NOTE - According to mfr. recommendation. (See MRB appendix 88.7 or MPG Appendix VI)	49-03 Check APU CHIP DETECTOR. NOTE - According to manufacturer's recommendation. (See MRB Appendix 8.7 or MPG Appendix VI).	24-13 Verify APU starter-generator GCU (GENERATOR CONTROL UNIT) operation	24-11 Visually check APU starter-generator QUICK ATTACH/DETACH QAD KIT generator condition. NOTE - At the time of starter generator restoration.	4 Day - Service Check	520-07 Verify operation of escape hatch locking mechanism. NOTE - Requires removal of escape hatches - 400FH or 2MO



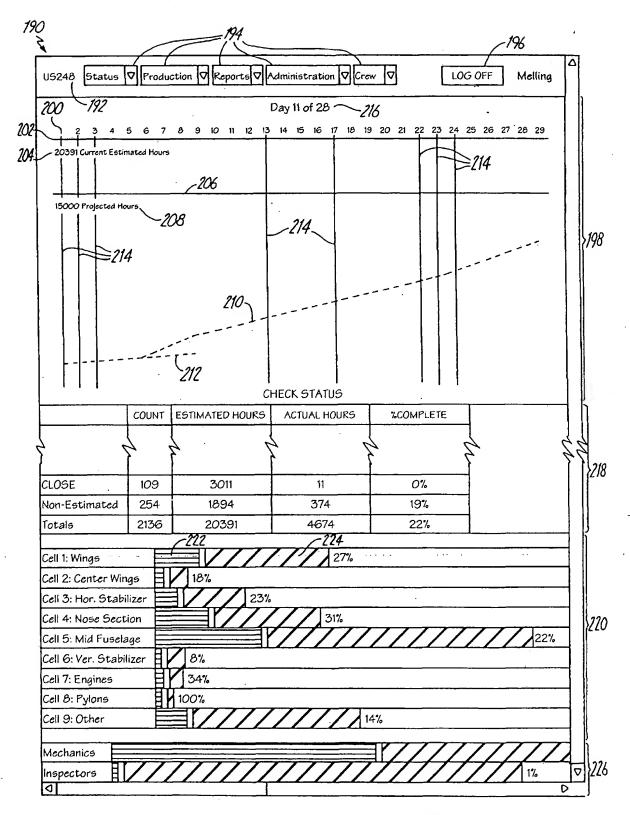


FIG. 9

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		sun mon tue wed thu fri sat sun 36 37 38 39 40 41 42 43 83 TD LE STAI98 (MRKD) **11015	O BASE METAL (MKD) "11020 BO	€ & INTO BASE METAL **11035		757			NCE AREA (MKD) **11075		E & INTO BASE METAL(MKD)				F16 to	<u>)</u>	
738	LOG OFF Melling	sun mon tue wed thu fri sat sun 29 30 31 32 33 34 35 36 E ENG. AT REMOVED PANEL LE STAIB3 1	WORN THRU ITS HARD CHROME & IN RADE FROM PY ON TO WING TIP **10	SY IS WORN THRU ITS HARD CHROM FROM CARRIAGE (MKD) **11040	KED **11045	GUIDE DEVICE "11055		TAS40 TO WING TP (MRKD) **11070	ED) IS WORN EXCESSIVELY AT ENTRA		SY IS WORN THRU ITS HARD CHROM	96011	<i>)</i>				
	Flow Chart (c) Administration (c) Crew (wed thu fri sat thu fri sat sun mon tue wed thu fri sun mon tue wed thu fri sat thu fri sat thu fri sat thu fri sat sat <td>ER BEARING ON #7 FLAP TRACK ASSY IS :D WING TEFLON COATING REQUIRES UPG</td> <td>INBD FLAP CARRIAGE ROLLER BEARING ON #7 FLAP TRACK ASSY IS WORN THRU ITS HARD CHROME & INTO BASE METAL ""11035 RUB STRIP ON OTBD EDGE OF #5 FLAP CARRIAGE IS DETACHED FROM CARRIAGE (MKD) ""11040</td> <td>.AS UNDERSIDE/AFT TEFLON AREAMAR</td> <td>IP LIGHT LENS HAS CONDENSATION INSIG #8 FLAP FAIRING IS CHAFED DEEPLY AT</td> <td>ES PAINT UPGRADE ""11060</td> <td>DEGRADED AT BELLY FS907. **11065 AMAGE BETWEEN FRT & REAR SPAR W/S</td> <td>ND OF O TBD RH FOREFLAP (WHEN STOW</td> <td>T UNDER PANEL 6301 IS TORN **11080</td> <td>LAP TRK FAIRING "11085 INBD FLAP CARRIAGE ROLLER BEARING ON #7.FLAP TRACK ASSY IS WORN THRU ITS HARD CHROME & INTO BASE METAL(MKD)</td> <td>IE 6" INB OF INB L/E FLAP ACT FLAGGED</td> <td>1AFED AREA **11100</td> <td>S</td> <td></td> <td></td> <td></td>	ER BEARING ON #7 FLAP TRACK ASSY IS :D WING TEFLON COATING REQUIRES UPG	INBD FLAP CARRIAGE ROLLER BEARING ON #7 FLAP TRACK ASSY IS WORN THRU ITS HARD CHROME & INTO BASE METAL ""11035 RUB STRIP ON OTBD EDGE OF #5 FLAP CARRIAGE IS DETACHED FROM CARRIAGE (MKD) ""11040	.AS UNDERSIDE/AFT TEFLON AREAMAR	IP LIGHT LENS HAS CONDENSATION INSIG #8 FLAP FAIRING IS CHAFED DEEPLY AT	ES PAINT UPGRADE ""11060	DEGRADED AT BELLY FS907. **11065 AMAGE BETWEEN FRT & REAR SPAR W/S	ND OF O TBD RH FOREFLAP (WHEN STOW	T UNDER PANEL 6301 IS TORN **11080	LAP TRK FAIRING "11085 INBD FLAP CARRIAGE ROLLER BEARING ON #7.FLAP TRACK ASSY IS WORN THRU ITS HARD CHROME & INTO BASE METAL(MKD)	IE 6" INB OF INB L/E FLAP ACT FLAGGED	1AFED AREA **11100	S			
23.	General O Production O Flow Chart	thu fri sat sun mon tue wed thu fri sat 12 13 14 15 16 17 18 19 20 21 J2·10-5726Q-002 R/WING UPPER SURFACE FRETTIN	J2-10-5726Q-025-FWD OTBD FLAP CARRIAGE ROLLER BEARING ON #7 FLAP TRACK ASSY IS WORN THRU ITS HARD CHROME & INTO BASE METAL (MKD) ""11020 J2-10-5726Q-003-RWING UPPER SURFACE LJE FIXED WING TEFLON COATING REQUIRES UPGRADE FROM PY ON TO WING TIP ""1060	J2-10-5726Q-024-UPPER AFT INBD FLAP CARRIAGE ROLLER BEARING ON #7 FLAP TRACK ASSY IS WORN THRU ITS HARD CHR. J2-10-5726Q-030-PHENOLIC RUB STRIP ON OTBD EDGE OF #5 FLAP CARRIAGE IS DETACHED FROM CARRIAGE (MKD) ***11040	J2-11-57502-023-0VERTIME #2 SEVERAL WORN AREAS UNDERSIDE/AFT TEFLON AREAMARKED **11045	J2-10-5726Q-005-R/WING UPPER SURFACE, WING TIP LIGHT LENS HAS CONDENSATION INSIDE ***:1050 J2-10-5726Q-029-LANDING LITE ACTUATING ROD IN #8 FLAP FAIRING IS CHAFED DEEPLY AT GUIDE DEVICE	J2-10-5726Q-037-RAWING LOWER SURFACE REQUIRES PAINT UPGRADE **11060	J2-50-53500-19-0VERTIME PROD SPLICE SEALANT DEGRADED AT BELLY F9907. **11065 J2-10-5726Q-006-R/WING UPPER SURFACE, HAIL DAMAGE BETWEEN FRT & REAR SPAR W/STA540 TO WINGTIP (MRKD) **11070	J2-10-5726Q-032-BEARING RECEPTACLE AT INBO END OF 0 TBD RH FOREFLAP (WHEN STOWED) IS WORN EXCESSIVELY AT ENTRANCE AREA (MKD) **11075	J2-11-57301-C-001-RUBBER BOOT ON BLEED AIR DUCT UNDER PANEL 6301 IS TORN **11080	J2-15-27500-INSP LT FLAP TRK FAIRING **11085 J2-10-5726Q-026 LOWER AFT INBD FLAP CARRIAGE ROI **11090	J2-10-5726Q-033-R WING CLAMP BROKEN - HYD LINE 6" INB OF INB LIE FLAP ACT FLAGGED ***11095	J2-11-57500-C-002-FIXED L/E ABOYE #1 ENG HAS CHAFED AREA "11100	J2-15-93864-WINGS-INSTL RLF-VLV/ARRESTR "11105			:
230 24	234 US248 Gener	240 11 12 13 242		1 J2-10-572 I J2-10-572	J2-11-575C	I J2-10-572.	- J2-10-572	I J2-50-538	I J2-10-5724	J2-11-573C	111 J2-10-5728)2-10-572k	J2-11-5750	11 12-15-938(

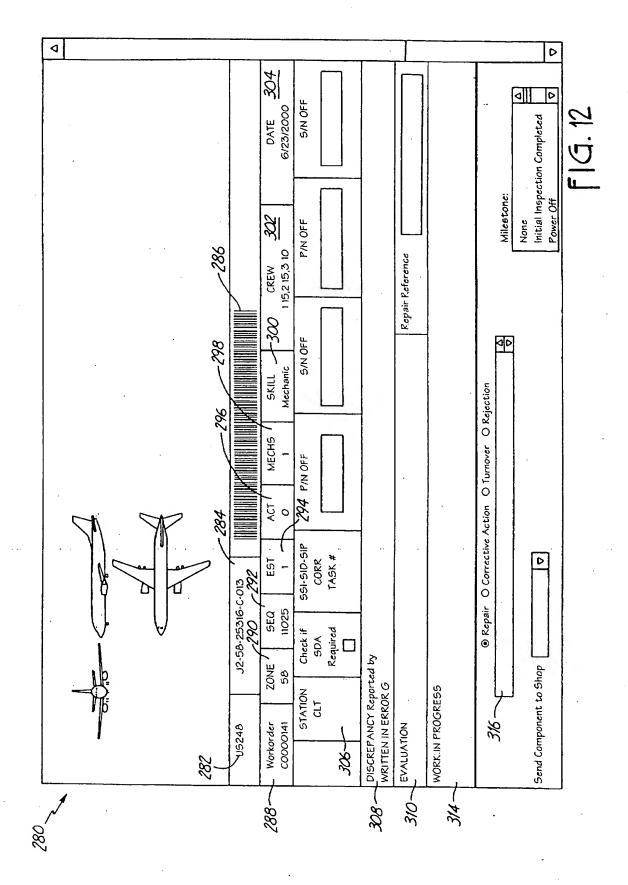
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	General	Tasks	7502-02	MultiTask Richter	J2-15-27500 INSP LT FLAP TR	MultiTask Richter	3864 W	MultiTask Richter	J2-42-52500-C INSP NOSE COMP AC DR (AD)	MultiTask Richter				
	-05248	Get All Tasks	J2-11-57502-023 #2 SEVERAL WORN AREAS UNDERSIDE/AFT TEFLON AREAMARKED	Mech 1 Hrs	J2-15-2	Mech 1 Hrs 0	J2-15-93864 WINGS-INSTL RLF-VLV/ARRESTR	Mech 1 Hrs	J2-42-E	Mech 1 Hrs				
250~	252		254 {	· /	007						-			

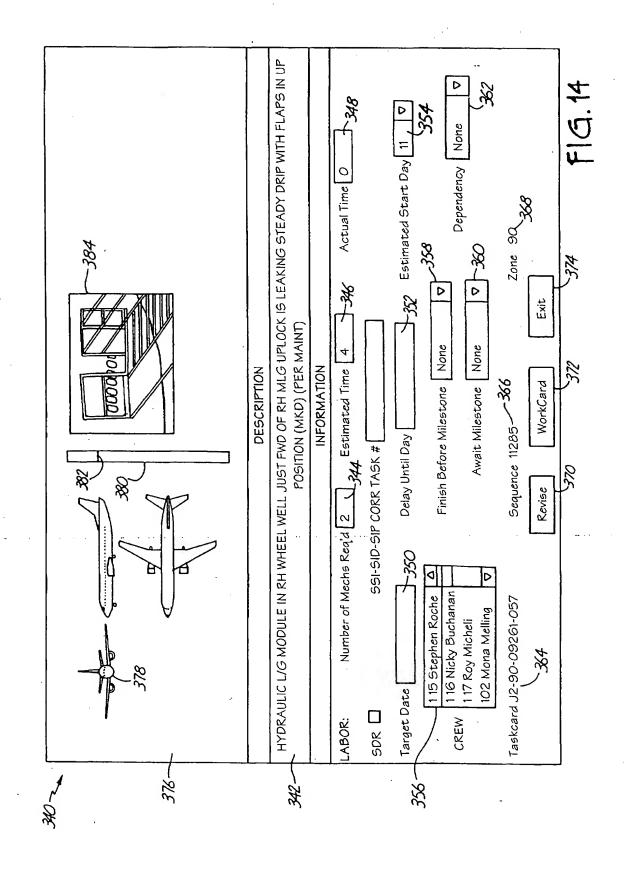
FIG. 11A

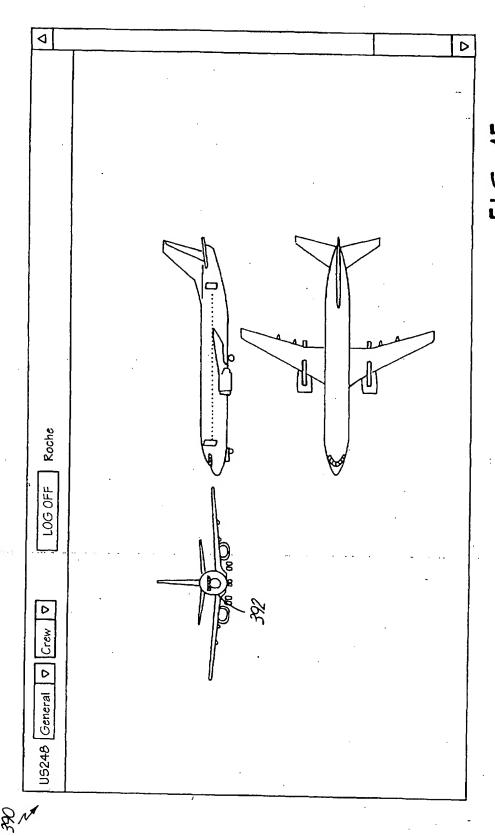
US248 General 🛡 Crew 🛡 LOG OFF Albin.	
Get All Tasks	-
CURRENT ASSIGNMENT	
J2-11-57502-023 #2 SEVERAL WORN AREAS UNDERSIDE/AFT TEFLON AREAMARKED M.Albin, Inspector	Work Card
J2-15-27500 INSP LT FLAP TRK FAIRING M. Albin, Inspector	Work Card
J2-15-93864 WINGS-INSTL RLF-VLV/ARRESTR M. Albin, Inspector, Callison	Work Card

FIG. 118

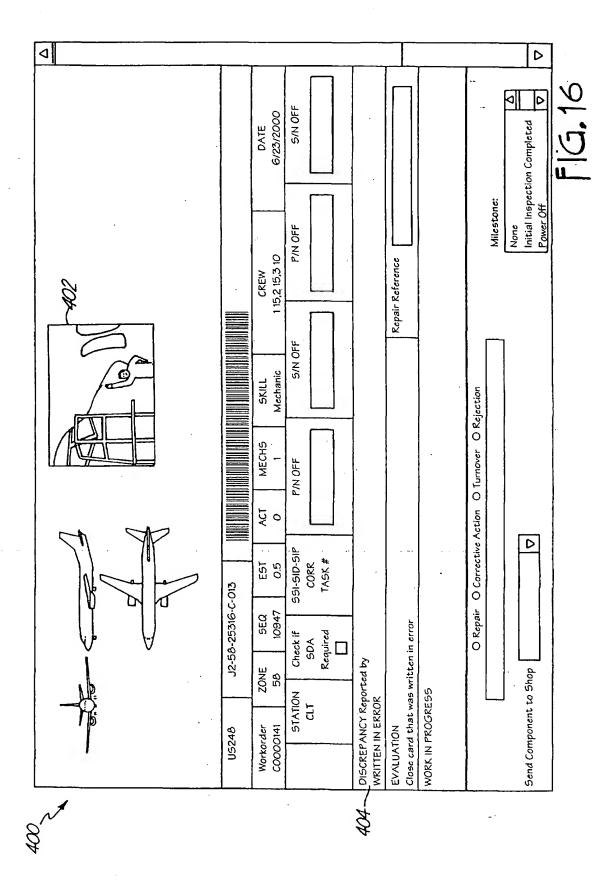


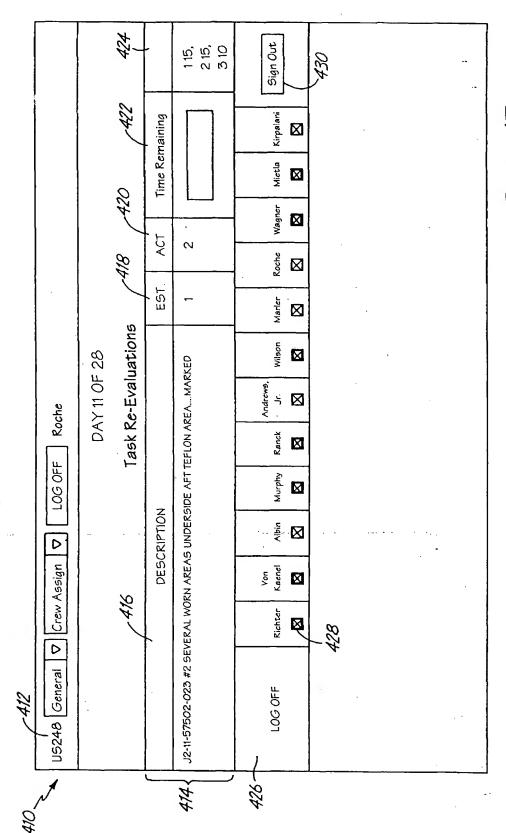
An inspection and test of the anti-collision strobe lights on each CRJ aircraft by Airlines. A ni inspection and test of the anti-collision strobe lights on each CRJ aircraft operated by Airlines. B Documentation of each aircraft anti-collision strobe light inspection and test will be made in the appropriate blocks in form #MOZZRJ (Ref. Appendix C.3), as follows: Enter the required aircraft information at the top of the form AMTI-COLLISION LIGHT
TASK CARD Task Card # J3-10-000C 328 330 Work Order # C000014 And test of the anti collision strobe lights on each CRJ aircraft operated by Airlin formed on a six month re-inspection basis no feach aircraft anti collision strobe light inspection and test will be made in the ocks in form #M022RJ (Ref. Appendix C.3), as follows: ANTI-COLLISION LIGHT ANTI-COLLISION LIGHT





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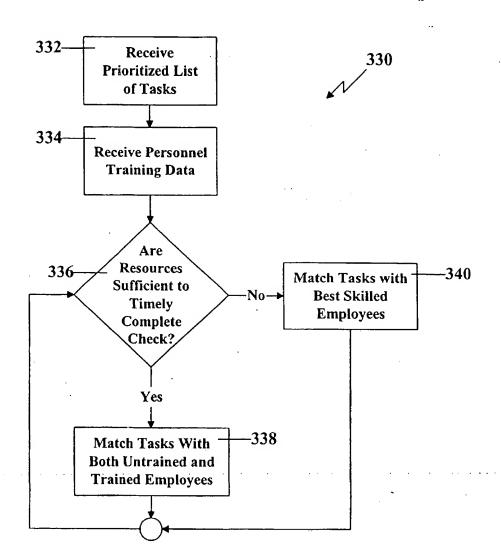


FIG.18 (AUTO ASSIGN)